

The Multiview Observatory for Solar Terrestrial Science (MOST)

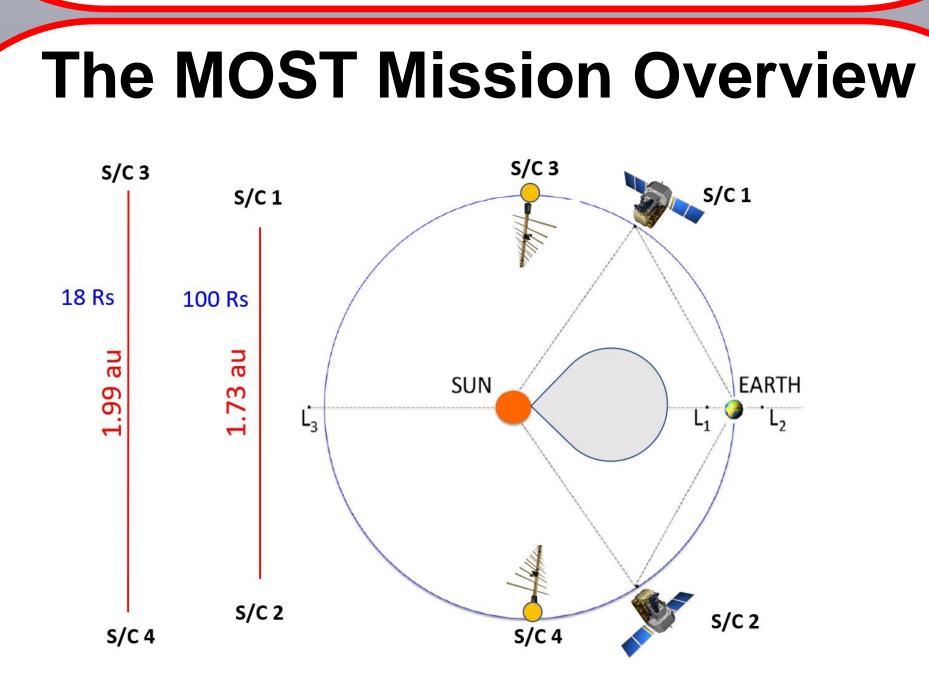
Introduction

The heliosphere is defined and structured by the magnetic field that emerges through the solar surface. To understand the magnetic coupling between the Sun and the 3-D heliosphere, we need to obtain the magnetic flux over the entire surface of the Sun, not just the 60-degree wedge observable from the Sun-Earth line. Simulations show that magnetic field measurement over a larger solar surface including the poles dramatically improve modeling of solar wind structures (Petrie et al. 2018; Pevtsov et al. 2020).

Solar events such as flares, jets, coronal mass ejections (CMEs), and solar energetic particles are all powered by solar magnetic energy released in the corona, yet we have a very limited set of magnetic field measurements: in the photosphere from the Sun-Earth line and in-situ observations in the solar wind.

The Multiview Observatory for Solar Terrestrial Science (MOST) will provide comprehensive imagery and time series data needed to understand the magnetic connection between the solar interior and the solar atmosphere.

MOST will build upon the successes of SOHO and STEREO with new views, instrumentation, and capabilities from Sun-Earth L4 and L5 and their vicinities.



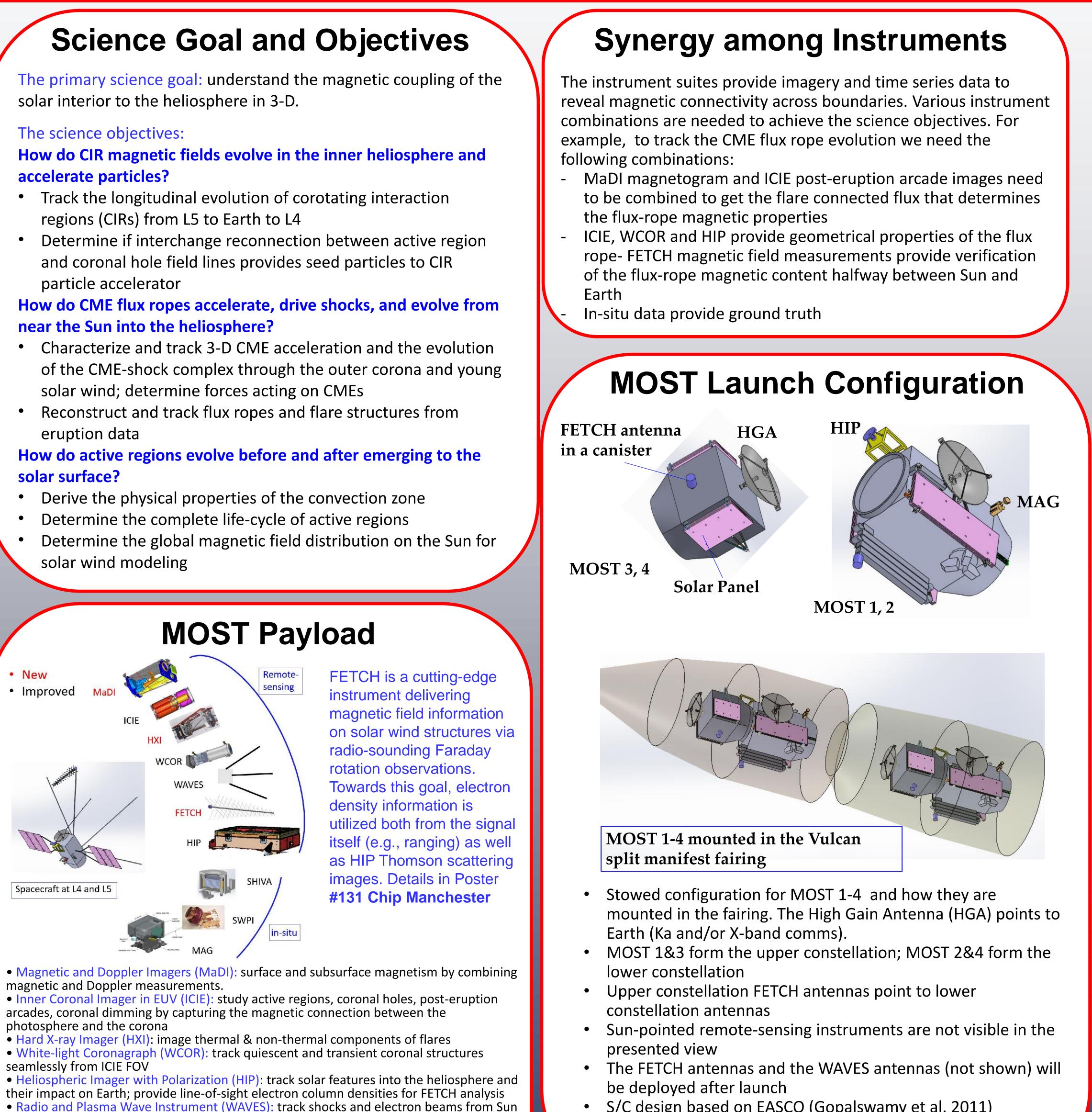
MOST is a four-spacecraft mission. Two large spacecraft each with seven remote-sensing and three in-situ instruments will be stationed at Sun-Earth L4 and L5. Two additional, smaller S/C drifting ahead of L4 and behind L5 will carry Faraday rotation radio package to measure the magnetic field in solar wind structures between the Sun and 0.5 au during the prime mission and the entire inner heliosphere in the extended mission. MOST will generate the following science data products: magnetograms, dopplergrams, EUV images, hard X-ray images, coronagraph images, heliospheric images, radio dynamic spectra and time series, Faraday rotation time series, time series of solar wind plasma parameters, solar wind magnetic field vectors, and solar energetic particle intensity and spectra.

References

Gopalswamy et al. 2011, JASTP 73, 658 Petrie, G J D et al. 2018, Solar Phys 293, 88 Pevtsov, A. et al. 2020, Space Weather, 18 (7), e02448 LASCO: Brueckner et al., 1995 WAVES: Bougeret et al., 1995; 2008 STEREO: Howard et al. 2008; Kaiser et al. 2008

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to 1 au • Faraday Effect Tracker of Coronal and Heliospheric Structures (FETCH): novel radio package to determine the magnetic field structure and evolution within 0.5 au

• Solar High-Energy Ion Velocity Analyzer (SHIVA): Energetic particle detector to determine the spectra of electrons and ions from H to Fe at multiple spatial locations and use energetic particles as tracers of magnetic connectivity

• Solar Wind Magnetometer (MAG): heliospheric magnetic structures at 1 au, CME and **CIR** evolution • Solar Wind Plasma Instrument (SWPI): to infer solar wind plasma structures at 1 au, CME

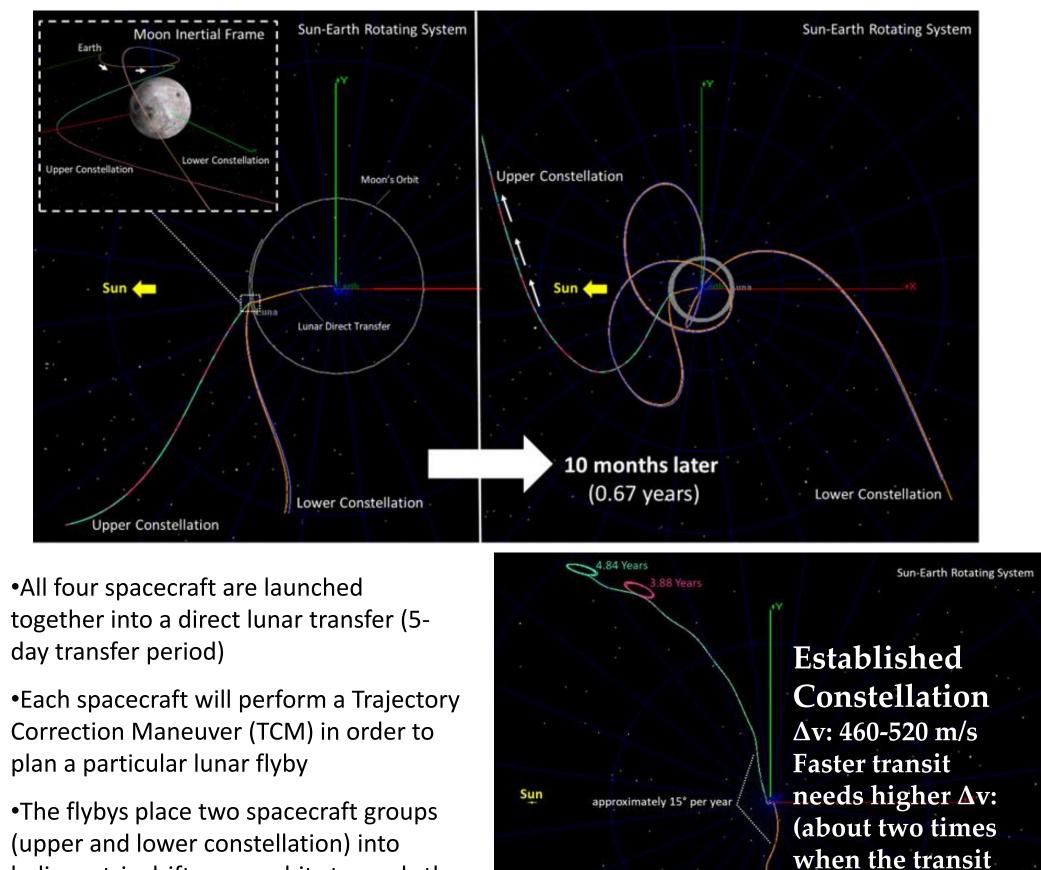
and CIR evolution

NASA/GSFC, CfA, NASA/MSFC, SWRI, NRAO, JPL, PSI, UMICH, UNH, NRL, UML, UCB, NSO, Lockheed, Stanford, CU, GMU

• S/C design based on EASCO (Gopalswamy et al. 2011)

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Lunar flyby performed to place the spacecraft into heliocentric orbits out to Sun-Earth L4 or L5 (similar to STEREO) High thrust (chemical propulsion) and low trust (ion propulsion) considered. **Dwell Phase:** Spacecraft remain at desired locations. Spacecraft can remain at desired locations with minimum station-keeping propellant. Drift Phase: MOST-1/2 stay at Sun-Earth L4/L5 indefinitely while MOST-3/4 begin an approximate 10-year drift campaign to drift back towards Earth and switch position.



L5 point

MOST is focused on two things: Observing subsurface magnetic flux concentrations ahead of emergence through the photosphere and understanding the global impact of the emerging flux from the inner corona out to 1 au. Such a comprehensive study needs a great observatory with a complete set of instruments. FETCH is a novel concept requiring the analysis of spacecraftto-spacecraft radio signals. Flight dynamics studies indicate that ion propulsion is a viable option. More trade studies will be performed between chemical and ion propulsion.



is shortened by 1

year)

Time since Launch

#55 SHINE 2022

Flight Dynamics

heliocentric drift-away orbits towards the L4 point (upper constellation), while the other two spacecraft (lower constellation) are directed towards the

•Low thrust option requires ~11 kg of mass consumption per S/C

Summary